

1 techniques: bonding wet lay-up sheets or laminates to the external faces of the elements to be strengthened
2 (Externally Bonded Reinforcement - EBR - technique) [4-7]; installing CFRP bars (circular, square or rectangular
3 cross section) into pre-cut slits opened on the concrete cover of the elements to strengthen (Near Surface Mounted -
4 NSM - technique) [8-11]. Due to the largest bond area and higher confinement provided by the surrounding
5 concrete, narrow strips of CFRP laminates of rectangular cross section, installed into thin slits and bonded to
6 concrete by an epoxy adhesive, are the most effective CFRP strengthening elements for the NSM technique [12].
7 Dias and Barros demonstrated by experimental research that NSM technique provides higher effectiveness than
8 EBR technique for the shear strengthening of rectangular cross sectional RC beams without steel stirrups [13] and T
9 cross sectional RC beams having a certain percentage of existing steel stirrups [14].

10 There are several reasons that justify the relevance of a study on the use of the NSM technique with CFRP
11 laminates for the shear strengthening of RC beams of low strength concrete: old RC structures were built with low
12 strength concrete; decrease of concrete strength due to several time dependent phenomena and environmental
13 conditions; experimental results of shear strengthening of RC beams using NSM technique with CFRP laminates
14 indicate that concrete has an important role in the effectiveness of this technique [14-15]. This last experimental
15 evidence was also obtained with a recent analytical/numerical model that involves the kinematic conditions of the
16 shear crack propagation in a NSM shear strengthened RC beam, as well as the concrete fracture and laminate-
17 adhesive-concrete bond main characteristics [16].

18 To appraise the possibility of the application of NSM CFRP laminates for the shear strengthening of T cross
19 sectional reinforced low strength concrete beams having a certain percentage of existing steel stirrups, an
20 experimental program was carried out. The average value of the concrete compressive strength at age of the beams
21 tests was 18.6 MPa. The experimental program is outlined and the specimens, materials and test set-up are
22 described. The results of the tests are presented and discussed and a number of conclusions are drawn. Taking the
23 results obtained in a previous experimental program [14-15], characterized by the same test set-up but using beams
24 of higher concrete strength (39.7 MPa instead of 18.6 MPa), the influence of the concrete mechanical properties in
25 the performance of the NSM technique with CFRP laminates for the shear strengthening of RC beams was assessed.

26 Additionally, the predictive performance of the formulation proposed by Nanni *et al.* [11] for the NSM shear
27 strengthening technique was appraised taking the results obtained in the present experimental program.

1 2. Experimental program

2 2.1 Beam prototypes

3 Fig. 1 presents the T cross section of the thirteen beams comprising the experimental program. The
4 reinforcement systems were designed to assure shear failure for all the tested beams. To localize the shear failure in
5 one only of the beam shear spans, a three point load configuration of distinct length of the beam shear spans was
6 selected, as shown in Fig. 1. The monitored beam span (L_i) is 2.5 times the effective depth of the beam cross section
7 ($L_i/d=2.5$), since, according to the available research [17], this is the minimum value with negligible arch effect. To
8 avoid shear failure in the L_r span, steel stirrups $\phi 6@75\text{mm}$ were applied in this span. The differences between the
9 tested beams are restricted to the shear reinforcement systems applied in the L_i beam span.

10 The experimental program is made up of three reference beams and two groups of NSM shear-strengthened
11 beams. The reference beams comprehend (see Fig. 1 and Fig. 2): one beam without any shear reinforcement (C-R
12 beam); one beam with steel stirrups $\phi 6@300\text{mm}$ (2S-R beam, with a percentage of stirrups, ρ_{sw} , of 0.10%); one
13 beam with steel stirrups $\phi 6@180\text{mm}$ (4S-R beam, with a percentage of stirrups, ρ_{sw} , of 0.17%). For the NSM
14 shear-strengthened beams, the first group is composed by five beams (2S-7LV, 2S-4LI45, 2S-7LI45, 2S-4LI60 and
15 2S-6LI60) presenting the percentage of stirrups as adopted in the 2S-R reference beam ($\rho_{sw} = 0.10\%$), and having
16 the CFRP shear strengthening arrangements indicated in Table 1 and Fig. 2. The second group also comprehends
17 five beams (4S-7LV, 4S-4LI45, 4S-7LI45, 4S-4LI60 and 4S-6LI60), having the percentage of stirrups used in the
18 4S-R reference beam ($\rho_{sw} = 0.17\%$), and the adopted strengthening configurations were the same applied in the
19 first group of beams (see Table 1 and Fig. 2). Three CFRP orientations (45° , 60° and 90°) were tested, and for
20 inclined laminates two CFRP percentages were adopted. For the both two groups the beams with the lower
21 percentage of laminates were designed in order to present similar maximum load, regardless of the orientation of the
22 laminates [14]. The same strategy was adopted on the design of the beams strengthened with the higher percentage
23 of laminates. Moreover, the two groups of beams were also conceived to study the influence of the amount of
24 existing steel stirrups on the effectiveness of the NSM shear strengthening technique.

25 According Fig. 1, the laminates were distributed along the AB line, where A represents beam support at its “test
26 side” and B is obtained assuming a load degradation at 45° .

27 The three point beam bending tests (Fig. 1) were carried out using a servo closed-loop control equipment, taking
28 the signal read in the displacement transducer (LVDT), placed at the loaded section, to control the test at a
29 deflection rate of 0.01 mm/second. To prevent brittle spalling of the concrete cover at the supports, the beam ends

1 were strengthened by confining the concrete with a two-directional cage of $\phi 6@65\text{mm}$ horizontal stirrups and
2 $\phi 12@50\text{mm}$ vertical stirrups $\phi 6\text{ mm}$ (Fig. 1). To overcome the difficulties to bend $\phi 32\text{ mm}$ longitudinal tensile bars,
3 their ends were welded to steel plates.

4 With the purpose of obtaining the strain variation along the two laminates that have the highest probability of
5 providing the largest contribution for the shear strengthening of the RC beam, four strain gauges (SG_L) were
6 bonded in each CFRP according to the arrangement represented in Fig. 3. Adopting the same principle, one steel
7 stirrup was monitored with three strain gauges (SG_S) installed according to the configuration represented in Fig. 3.
8 The location of the monitored laminates and stirrups in the tested beams is represented in Fig. 2.

9 10 *2.2 Material properties*

11 The concrete compressive strength was evaluated at 28 days and at the age of the beam tests, carrying out direct
12 compression tests with cylinders of 150 mm diameter and 300 mm height, according to EN 206-1 [18]. In the tested
13 beams, high bond steel bars of 6, 12, 16 and 32 mm diameter were used. The values of their main tensile properties
14 were obtained from uniaxial tensile tests performed according to the recommendations of EN 10002-1 [19]. The
15 tensile properties of the CFK 150/2000 S&P laminates, with a cross section area of $1.4 \times 9.5\text{ mm}^2$, were
16 characterized by uniaxial tensile tests carried out according to ISO 527-5 [20]. Table 2 includes the average values
17 obtained from these experimental programs. The MBrace Resin 220 [21] epoxy adhesive was used to bond the
18 laminates to the concrete.

19 20 *2.3 Strengthening technique*

21 To apply the precured CFRP laminates using NSM technique, the following procedures were executed: 1) using
22 a diamond cutter, slits of about 4-5 mm width and 12-15 mm depth were opened on the concrete cover (of about
23 22 mm thickness) of the lateral faces of the beam web, according to the pre-defined arrangement for the laminates
24 (the laminates were not anchored to the beam flange; they were restricted to the beam web); 2) the slits were cleaned
25 by compressed air; 3) the laminates were cut with the desirable length and cleaned with acetone; 4) the epoxy
26 adhesive was produced according to the supplier recommendations; 5) the slits were filled with the adhesive; 6) a
27 layer of adhesive was applied on the faces of the laminates; and 7) the laminates were inserted into the slits and
28 adhesive in excess was removed.

29 To guarantee a proper curing of the adhesive, at least one week passed between the beam strengthening
30 operations and the beam test.

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2 3. Results

3 3.1 Load carrying capacity of the tested beams

4 The relationship between the applied force and the deflection at the loaded section (u_{LS}) for the tested beams is
5 represented in left part of Fig. 4 and Fig. 5, respectively, for the lower and higher percentage of steel stirrups. As
6 Fig. 4b shows, ΔF represents the increase of the load provided by a shear strengthening system, while F^{ref} is the
7 corresponding load capacity of the reference beam. For deflections greater than the corresponding to the formation
8 of the first shear crack in the reference beam, the $\Delta F/F^{ref}$ ratio was evaluated, and the $\Delta F/F^{ref}$ vs u_{LS}
9 relationship is depicted in the right part of Fig. 4 (the reference beam is 2S-R) and Fig. 5 (the reference beam is 4S-
10 R). For the tested beams, the $(\Delta F/F^{ref})_{max}$ value and the corresponding deflection at the loaded section,
11 $u_{(\Delta F/F^{2S-R})_{max}}$, are indicated in Table 3. Assuming that F_{max} and F_{max}^{ref} are the load carrying capacities (maximum
12 force) of strengthened and reference beams (2S-R or 4S-R) the values of $\Delta F_{max}/F_{max}^{ref}$ were evaluated
13 ($\Delta F_{max} = F_{max} - F_{max}^{ref}$). These values and the deflection at loaded section associated to F_{max} ($u_{F_{max}}$) are included in
14 Table 3.

15 Fig. 4 and Fig. 5 and the results included in Table 3 show that, for deflections higher than the one corresponding
16 to the formation of the first shear crack in the reference beams ($u_{LS} = 1.29$ mm for 2S-R (Fig. 4) and $u_{LS} = 1.26$ mm
17 for 4S-R (Fig. 5)), the adopted CFRP configurations provided an increase in the beam's load carrying capacity. This
18 reveals that the CFRP laminates bridging the surfaces of the shear crack offer resistance, mainly, to crack opening,
19 resulting a smaller degradation of the shear stress transfer between the faces of the crack due to aggregate interlock
20 effect. Therefore, for deflections above the deflection corresponding to the formation of the shear crack in the
21 reference beams, an increase of the beam's stiffness is observed in the shear strengthened beams. The crack opening
22 resisting mechanisms provided by the crack bridging laminates also contribute to increase the load at which stirrups
23 enter in their plastic phase.

24 The strengthening arrangements with the lower percentage of CFRP (only beams with inclined laminates) had
25 the following increments in terms of beam load carrying capacity ($\Delta F_{max}/F_{max}^{ref}$): 24.9% and 24.3% for the beams
26 strengthened with laminates at 45° (2S-4LI45 beam) and laminates at 60° (2S-4LI60 beam), respectively, for the
27 beams with two steel stirrups in the smaller beam shear span; 14.3% and 13.8% for the beams strengthened with

1 laminates at 45° (4S-4LI45 beam) and laminates at 60° (4S-4LI60 beam), respectively, for the beams with four steel
 2 stirrups in the smaller beam shear span. In terms of $\left(\Delta F / F^{ref}\right)_{max}$, only the beams with laminates at 60° presented
 3 higher value than the one obtained for $\Delta F_{max} / F_{max}^{ref}$ (27.4% for 2S-4LI60 beam and 19.6% for 4S-4LI60).
 4 Laminates at 45° and 60° had similar performance in terms of maximum load. The beams with laminates at 60° had
 5 better performance than the beams with laminates at 45° in terms of stiffness.

6 For the beams shear strengthened with the higher CFRP percentage, the strengthening configurations with
 7 inclined laminates were more effective than CFRP arrangement at 90°. Vertical laminates, laminates at 45° and
 8 laminates at 60° assured an increase of $\Delta F_{max} / F_{max}^{ref}$: 20.8%, 35.3% and 31.4%, respectively, for the beams with
 9 two steel stirrups in the smaller beam shear span; and 3.8%, 17.3% and 19.3%, respectively, for the beams with four
 10 steel stirrups in the smaller beam shear span. For the higher percentage of the CFRP the values of $\left(\Delta F / F^{2S-R}\right)_{max}$
 11 of the beams 2S-7LV (26.1%), 2S-7LI45 (38.5%), 4S-7LV (6.7%), 4S-7LI45 (19.3%) and 4S-6LI60 (20.9%) were
 12 higher than the values of $\Delta F_{max} / F_{max}^{ref}$. For the arrangements with laminates at 45° and 60° the values of
 13 $\left(\Delta F / F^{ref}\right)_{max}$ and $\Delta F_{max} / F_{max}^{ref}$ increased with the percentage of CFRP.

14 The comparison between Fig. 4 and 5 and the results included in Table 3 show that the amount of existing steel
 15 stirrups plays a very important role on the effectiveness of the NSM shear strengthening technique. In fact, the
 16 effectiveness of the CFRP was higher in the beams with the lower percentage of steel stirrups analyzed. According
 17 to the experimental results, for an increase from 0.1% to 0.17% in the percentage of steel stirrups in the L_i beam
 18 span, the NSM strengthening effectiveness decreased in about 55% (see Fig. 6 - the value regarding the
 19 configuration with vertical laminates was excluded). For the solutions with inclined laminates, the average value of
 20 $\Delta F_{max} / F_{max}^{ref}$ varies from 29.0% to 16.2% when the amount of stirrups changes from $\phi 6 @ 300$ mm ($\rho_{sw} = 0.10\%$) to
 21 $\phi 6 @ 180$ mm ($\rho_{sw} = 0.17\%$), meaning that the interaction with stirrups plays a detrimental effect on the effectiveness
 22 of the NSM strengthening system. This fact was also observed by Dias and Barros [15] when the five NSM
 23 arrangements adopted in the present experimental program were applied in identical beams with a concrete of higher
 24 compressive strength (in this case for an increase from 0.1% to 0.17% in the percentage of steel stirrups in the L_i
 25 beam span, the NSM strengthening effectiveness decreased in about 70% - see section 4). This is an important
 26 aspect due to the fact that existing RC beams requiring shear strengthening intervention often have a certain
 27 percentage of steel stirrups. It emerges that a formulation for the prediction of the NSM shear strengthening
 28 contribution cannot neglect the percentage of existing steel stirrups.

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3.2 Failure modes

As was expected, all the tested beams failed in shear (Fig. 7). In this figure, the steel stirrups of the smaller beam shear span are indicated by vertical lines, and the circles indicate the zone where stirrups have ruptured.

When the maximum load of the C-R beam was attained the shear failure crack widened abruptly. The maximum load of the 2S-R and 4S-R beams was attained when one stirrup crossing the shear failure crack had ruptured.

In the NSM beams strengthened with the lowest percentage of CFRP the laminates failed by “debonding”. However, in the present context “debonding” should not be assumed as a pure debonding failure mode of the laminate, since along its bond length, parts of concrete were adhered to the laminate, indicating that failure includes debond and concrete fracture, which is in agreement with the principles of the model of Bianco *et al.* [16]. In the beams with the highest percentage of CFRP a group effect takes place and the concrete cover delamination (premature detachment of a concrete layer that includes the laminates) is found to be the critical failure mode. This indicates that the efficacy of NSM shear strengthening might be limited by laminate spacing. Fig. 8 clarifies the influence of CFRP percentage in the failure modes of the beams.

The global analysis of failure modes of the tested beams with CFRP indicates that the efficacy of NSM technique for the shear strengthening of RC beams using CFRP laminates is dependent of the concrete strength and increases with the quality of this material. In section 4 of the present paper it is possible to verify this evidence by experimental results.

3.3 Strains in the CFRP and steel stirrups

The maximum strain recorded in the laminates up to the maximum load of the beams (ϵ_{CFRP}^{max} - see Table 3) ranged from 0.54% (beam 4S-6LI60) and 0.94% (beam 2S-4LI45). In terms of the average strain of the maximum strain values registered in the monitored laminates (two per beam) ($(\epsilon_{CFRP}^{max})_{med}$ - see Table 3), the variation was between 0.41% in the 4S-7LV and 0.88% in the 4S-4LI45. For all the tested NSM beams the average value of ϵ_{CFRP}^{max} and $(\epsilon_{CFRP}^{max})_{med}$ was 0.72% and 0.62%, respectively. In terms of CFRP orientation, the average value of the maximum strain (ϵ_{CFRP}^{max}) was 0.81%, 0.67% and 0.64% for the beams with laminates at 45°, 60° and 90°, respectively. These values ranged from 39% to 50% of the CFRP ultimate rupture strain ($\epsilon_{fu} = 1.63\%$ - see Table 2).

To illustrate a representative strain variation in monitored laminates and stirrup during the beam loading process, the strain values for distinct load levels of the 4S-4LI45 beam are indicated in Table 4, from which it can be

1 observed that the maximum strain value was 0.84‰ (SG_L2 and SG_L3) in the CFRP A and 0.93‰ (SG_L2) in the
2 CFRP B. The strain values of the monitored stirrup of the 4S-4LI45 and 4S-R beams for four loads levels, included
3 in Table 4 (the values of the 4S-R beam are in the round brackets), show that the steel stirrup was more strained in
4 the reference beam than in the strengthened beam. The CFRP laminates bridging the faces of the shear failure crack
5 offer some resistance to the crack opening. This mechanism also decreases the loss of the concrete aggregate
6 interlock contribution for the shear resistance that occurs with the crack opening of the shear failure crack. Due to
7 these effects, the strains on the steel stirrups of the CFRP strengthened beams were lower than the strains recorded
8 in the reference beam, at equal load levels applied to the beams.

9 Fig. 9 represents the variation of the strains on the monitored laminates (see Table 4) during the loading process
10 up to the maximum load of the 4S-4LI45 beam. It is observed that the curves feature two phases. In the initial stage
11 of loading, the CFRP did not contribute to the load-carrying capacity of the beam. In the second stage, the laminate
12 began to strain due the formation of a crack that crossed the laminate. The strain in the laminate continued to
13 increase with the increase of the load up to the maximum force of the beam (the exception was the SG_L1 in the
14 CFRP A). Fig. 9 evidences that both laminates were crossed by more than one crack. In the CFRP A, SG_L2 and
15 SG_L3 were closer to the shear crack zone than the other strain gauges, while in the CFRP B this happened to the
16 SG_L2. Therefore the strain variation depends significantly on the relative position between SGs and the formed
17 cracks. Fig. 10 represents the variation of the strains on the monitored stirrup during the loading process up to the
18 maximum load of the 4S-4LI45 beam, being possible to observe the presence of the same two phases already
19 identified in laminates. In the last phase of the strain variation in the steel stirrup of the 4S-4LI45 beam the strain
20 registered in SG_S decreased with the increase of the load. This was caused by the formation of the shear failure
21 crack that did not cross this steel stirrup (Fig. 7).

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23 **4. Influence of the concrete strength on the effectiveness of the NSM technique**

24 In the present section the influence of concrete strength on the efficacy of the NSM technique is assessed
25 comparing the results of the experimental program described in previous sections with the results of other
26 experimental program [14, 15] dealing with beams of higher concrete strength. In this experimental program the five
27 arrangements of NSM CFRP laminates presented in Fig. 2 were also used but the beams were manufactured with a
28 concrete of a compressive strength at the age of beam tests of 39.7 MPa ($f_{cm} = 39.7$ MPa instead of $f_{cm} = 18.6$ MPa -
29 see Table 2). The remaining characteristics were the same in both experimental programs.

1 The values of the $\Delta F_{max}/F_{max}^{ref}$ ratio, $u_{F_{max}}/u_{F_{max}}^{ref}$ ratio ($u_{F_{max}}^{ref}$ is the the deflection at loaded section
2 corresponding to F_{max}^{ref}) and the maximum strain recorded in the laminates (ε_{CFRP}^{max}) for the beams of the above
3 mentioned experimental programs are included in Table 5. The $\Delta F_{max}/F_{max}^{ref}$ ratio is also represented in Fig. 11, being
4 visible the increase of the NSM effectiveness with the increase of the concrete strength. According to the values into
5 Table 5, the average value of the $\Delta F_{max}/F_{max}^{ref}$ ratio for NSM arrangements adopted in beams with higher and lower
6 concrete compressive strength was 26.5% and 22.4%, respectively (the values regarding the 4S-4LV shear
7 strengthening configuration was excluded for this analysis). The better performance of the CFRP when applied in the
8 beams with $f_{cm} = 39.7$ MPa was also observed in terms of the $u_{F_{max}}/u_{F_{max}}^{ref}$ ratio. The average value of the
9 $u_{F_{max}}/u_{F_{max}}^{ref}$ ratio for NSM arrangements adopted in beams with higher and lower concrete compressive strength
10 was 22.7% and 6.9%, respectively (the values regarding the 4S-4LV shear strengthening configuration was excluded
11 for this analysis). Furthermore, in the group of beams of higher concrete strength larger maximum strain values were
12 registered, which indicates that the laminates are more effectively mobilized as higher is the strength of the concrete.
13 The average value of the maximum strain recorded in the laminates up to the maximum load of the beams (ε_{CFRP}^{max})
14 was 0.9% for beams with $f_{cm} = 39.7$ MPa and 0.72% for beams with $f_{cm} = 18.6$ MPa.

15 The typical failure modes obtained in the series of NSM beams of distinct concrete strength class are characterized
16 by i) the fracture of a concrete volume surrounding the bond length of the CFRP laminate; ii) a mix mode composed
17 of concrete fracture followed by debond. In fact, after failure, a certain volume of concrete is bonded to the CFRP
18 laminate, and the propensity for the occurrence of the failure mode i) seems to be as high as lower is the concrete
19 strength class, as proved by Bianco et al. [16] the both experimental programs were similar (see section 3.2). However,
20 due the higher concrete strength, the decrease of the effective bond length, in consequence of the fracture failure of
21 concrete surrounding a laminate crossing the critical diagonal crack, was not so pronounced as occurred in the beams
22 of lower concrete strength class. In this case, the tensile strength of the concrete surrounding the CFRP bond length
23 is the main governing parameter of the NSM shear strengthening effectiveness. For beams of concrete of higher
24 strength class, concrete fracture is, in general, the first failure occurrence, leading to a reduction of the effective bond
25 length of the CFRP, thereby the aforementioned mix mode is the typical failure mechanism. In consequence, the
26 higher is the tensile strength of the concrete surrounding the laminates, the larger is the contribution of the NSM
27 CFRP laminates for the shear resistance of RC beams.

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5. Comparison of experimental and analytical results

Taking the results obtained in the tested beams strengthened with NSM technique, the predictive performance of the analytical formulation proposed by Nanni *et al.* [11] was appraised. According this formulation, the term within the square brackets of the equation (1) is the force resulting from the tensile stress in the NSM FRP elements crossing the shear failure crack. The vertical projection of this force is the contribution of the FRP to the shear resistance of the beam (V_f),

$$V_f = [4 \cdot (a_f + b_f) \cdot \tau_b \cdot L_{tot\ min}] \cdot \sin \theta_f \quad (1)$$

being τ_b the average bond stress of the FRP elements intercepted by the shear failure crack, and $L_{tot\ min}$ is obtained from (see Fig. 12),

$$L_{tot\ min} = \sum_i L_i \quad (2)$$

where L_i represents the length of each single NSM laminate intercepted by a 45-degree shear crack, determined from,

$$L_i = \begin{cases} \min \left(\frac{s_f}{\cos \theta_f + \sin \theta_f} i; l_{max} \right) & i = 1, \dots, \frac{N}{2} \\ \min \left(l_{net} - \frac{s_f}{\cos \theta_f + \sin \theta_f} i; l_{max} \right) & i = \frac{N}{2} + 1, \dots, N \end{cases} \quad (3)$$

$L_{tot\ min}$ corresponds to an arrangement of the FRP reinforcements crossing the shear failure crack that leads to the minimum of the $\sum_i L_i$. In (3) l_{net} is defined as,

$$l_{net} = l_b - \frac{2c}{\sin \theta_f} \quad (4)$$

which represents the net length of a FRP laminate, as shown in Fig. 12, to account for cracking of the concrete cover and installation tolerances. In (4), l_b is the actual length of a FRP laminate and c is the concrete clear cover thickness.

The first limitation of (3) takes into account bond as the controlling failure mechanism, and represents the minimum effective length of a FRP laminate intercepted by a shear crack as a function of the term N ,

$$N = \frac{l_{eff} (1 + \cot \theta_f)}{s_f} \quad (5)$$

1 where N is rounded off to the lowest integer (e.g., $N=5.7 \Rightarrow N = 5$), and l_{eff} represents the length of the vertical
 2 projection of l_{net} as shown in Fig. 12,

$$l_{eff} = l_b \sin \theta_f - 2c \quad (6)$$

3 The second limitation in (3), $L_i = l_{max}$, results from the force equilibrium condition, taking an upper bound value for
 4 the effective strain, \mathcal{E}_{fe} (see Fig. 13),

$$l_{max} = \frac{\mathcal{E}_{fe}}{2} \cdot \frac{a_f \cdot b_f}{a_f + b_f} \cdot \frac{E_f}{\tau_b} \quad (7)$$

5 The design contribution of the NSM laminates to the shear resistance of the beams is defined as,

$$V_{fd} = \phi \psi_f V_f \quad (8)$$

6 being ϕ the strength-reduction factor indicated by ACI [22] that, for shear strengthening of concrete elements has a
 7 value of 0.85. For the additional reduction factor ψ_f , the value of 0.85 recommended by ACI [1] for EBR technique
 8 (bonded face schemes) was adopted.

9 Adopting for \mathcal{E}_{fe} and τ_b the values recommended by Barros and Dias [13], respectively, 0.59% and 16.1 MPa,
 10 assuming for the elastic modulus of the laminate the average value recorded in the experimental program of the
 11 present work (174.3 GPa), the values of the contribution of the NSM laminates for the shear strengthening of the
 12 tested beams (V_{fd}^{ana}), included in Table 6, are compared to those registered experimentally (V_f^{exp}). These values were
 13 obtained by subtracting the shear resistance of the reference beam (2S-R beam or 4S-R beam) from the shear
 14 resistance of the NSM strengthened beam. Apart the too abnormal low contribution of the laminates registered in the
 15 tested 4S-7LV beam, the values of V_f^{exp} / V_{fd}^{ana} are very closer or higher than one (safety condition), and decrease
 16 with the increase of the CFRP percentage. An average value of about 1.47 for V_f^{exp} / V_{fd}^{ana} was obtained. Regarding
 17 the influence of the percentage of steel stirrups, the average value of V_f^{exp} / V_{fd}^{ana} was 1.59 and 1.32 for beams with
 18 ρ_{sw} equal to 0.10% and 0.17%, respectively. This difference can be justified by the fact the analytical formulation
 19 proposed by Nanni *et al.* [11] does not consider the detrimental effect of the influence of the percentage of transverse
 20 steel stirrups in the contribution of the NSM laminates.

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22 6. Conclusions

23 The effectiveness of the NSM technique for the shear strengthening of T cross section RC beams of low concrete

1 strength was analysed by carrying out an experimental program. This effectiveness was appraised by assessing the
2 contribution provided by the distinct CFRP shear strengthening arrangements in terms of load carrying capacity,
3 stiffness of the response of the beams after the formation of the shear failure crack in the reference beam, maximum
4 strains measured in the CFRP laminates, and failure modes. The influence of the percentage and inclination of the
5 laminates and the percentage of existing steel stirrups was also evaluated.

6 From the obtained results it can be concluded that the shear strengthening technique is still effective in RC
7 beams of an average concrete compressive strength of 18.6 MPa at the age of the beams tests, which can be
8 considered as the lowest concrete strength class for structural purposes. The CFRP shear strengthening
9 configurations provided an increase not only in terms of maximum load, but also in terms of load carrying capacity
10 after shear crack formation. The concrete strength has, however, an important role on the effectiveness of the NSM
11 shear strengthening technique, since this effectiveness decreases with the decrease of the concrete strength. In fact,
12 when the same NSM CFRP laminates arrangements were applied in a group of beams of concrete compressive
13 strength (f_{cm}) equal to 39.7 MPa and in another group of beams of $f_{cm}=18.6$ MPa, the CFRP laminates were more
14 effective in the former beams. In the beams of lower concrete strength class a predominant fracture of concrete
15 surrounding the bond length of the laminates has occurred, while in the series of beams of larger concrete strength
16 class a mix mode composed by concrete fracture followed by debond was the typical failure mode. The mix mode is
17 generally associated to a higher effectiveness of the NSM CFRP laminates for the shear resistance of RC beams than
18 the concrete fracture.

19 Inclined laminates were more effective than vertical laminates and an increase of the percentage of laminates led
20 to an increase of the shear capacity of the beams. A detrimental effect of the increase of the percentage of the
21 existing steel stirrups exists in terms of the effectiveness of the NSM technique for the shear resistance of RC
22 beams.

23 The formulation proposed by Nanni *et al.* for the NSM shear strengthening technique was applied to the tested
24 beams of low concrete strength. In general, this formulation provided safe and acceptable estimates for the
25 contribution of the NSM shear strengthening systems (the predicted values of the CFRP contribution for the shear
26 resistance were 75% of the results registered experimentally).

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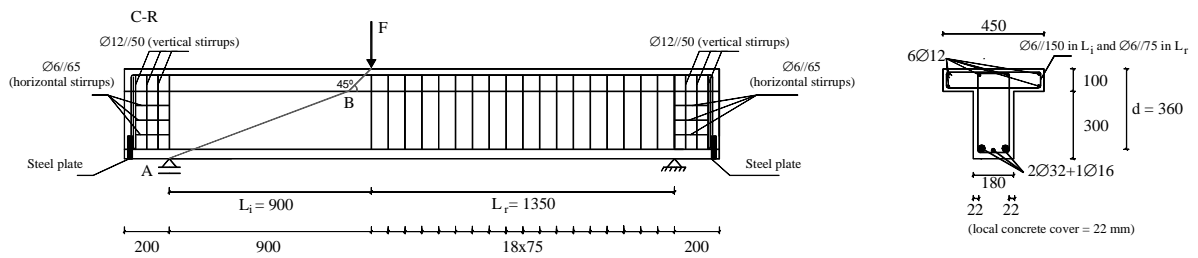
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7 **Fig. 1 - Tested beams: geometry, steel reinforcements applied in all beams (dimensions in mm)**

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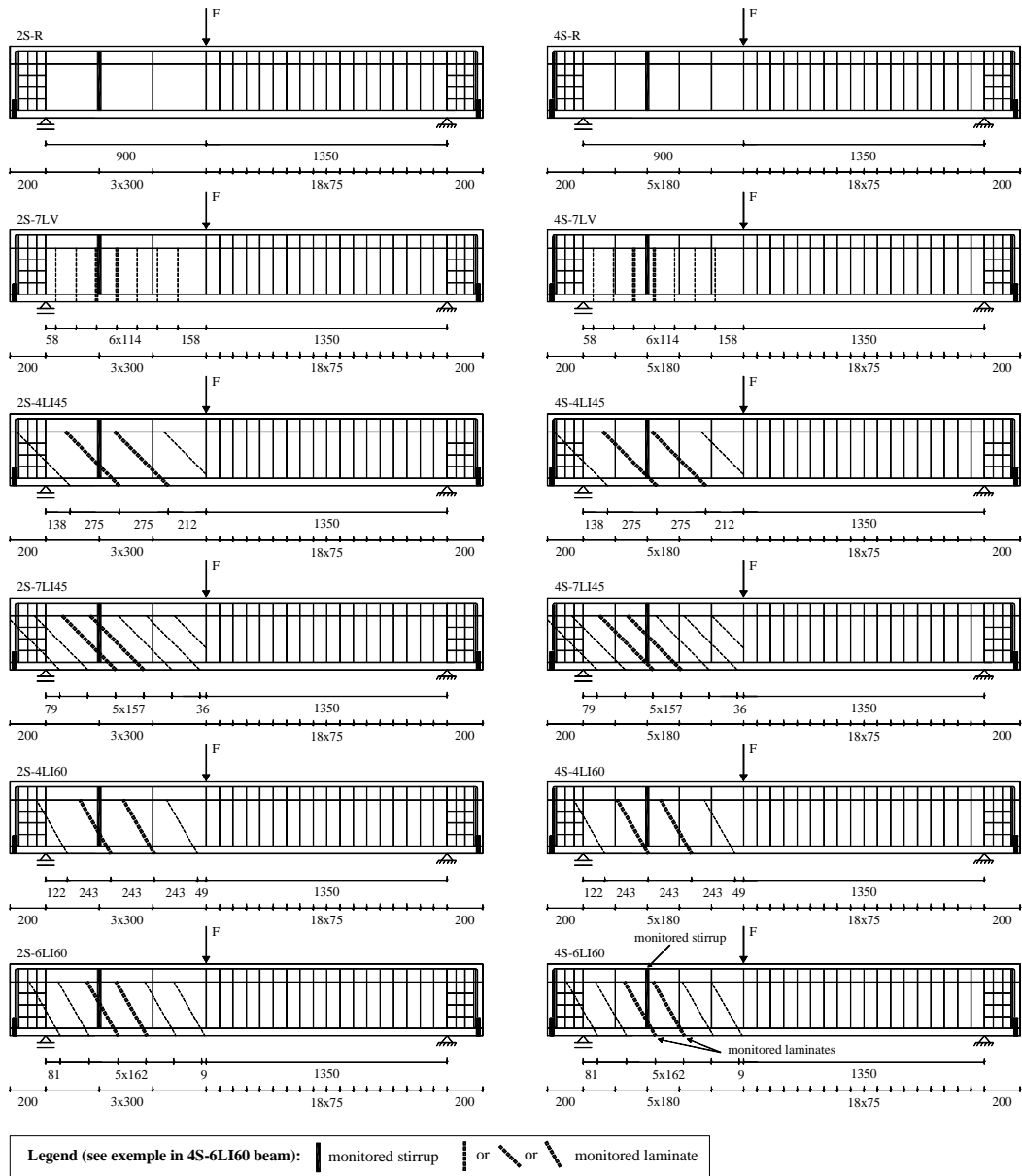


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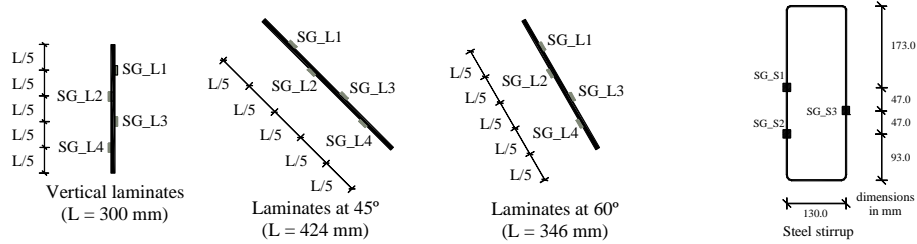
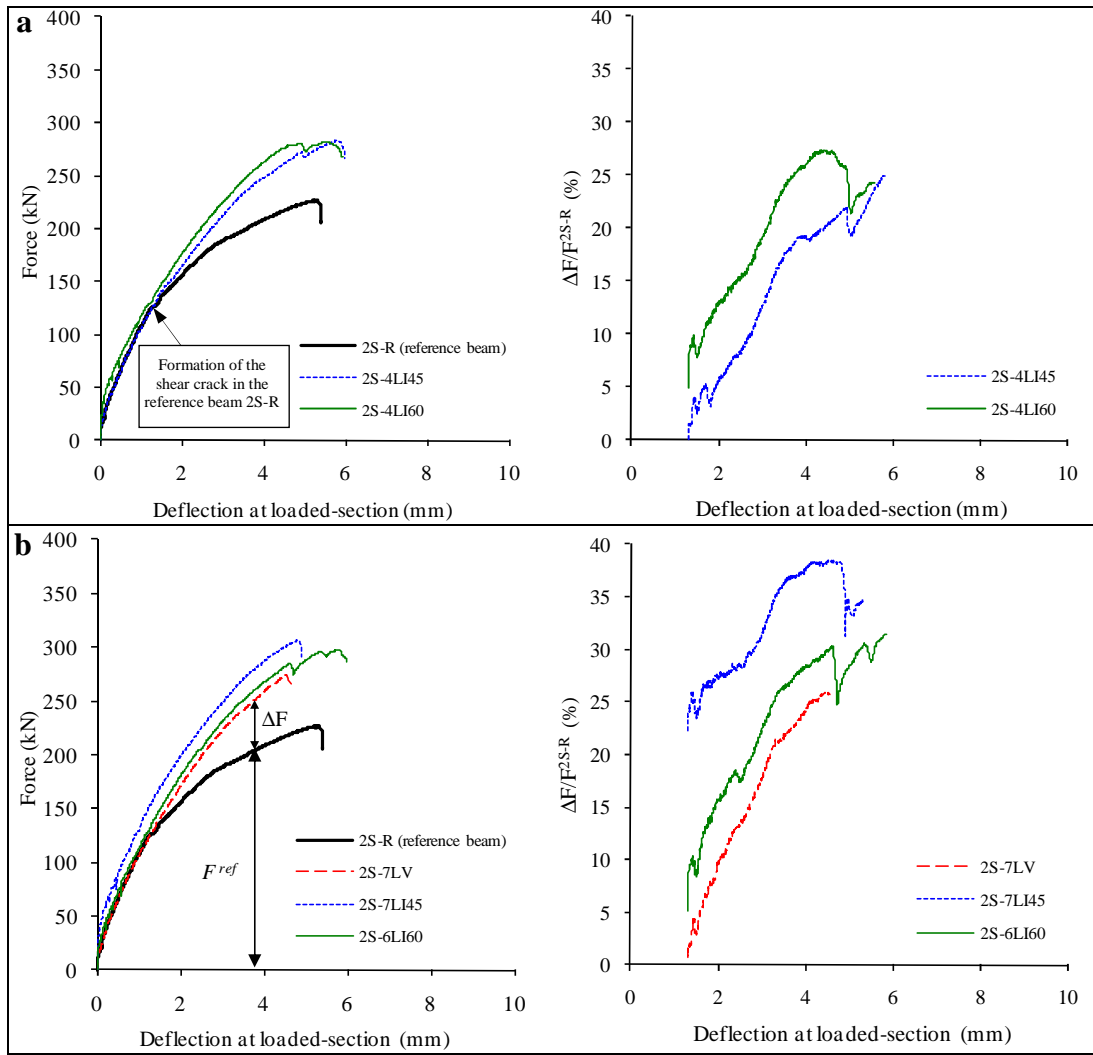
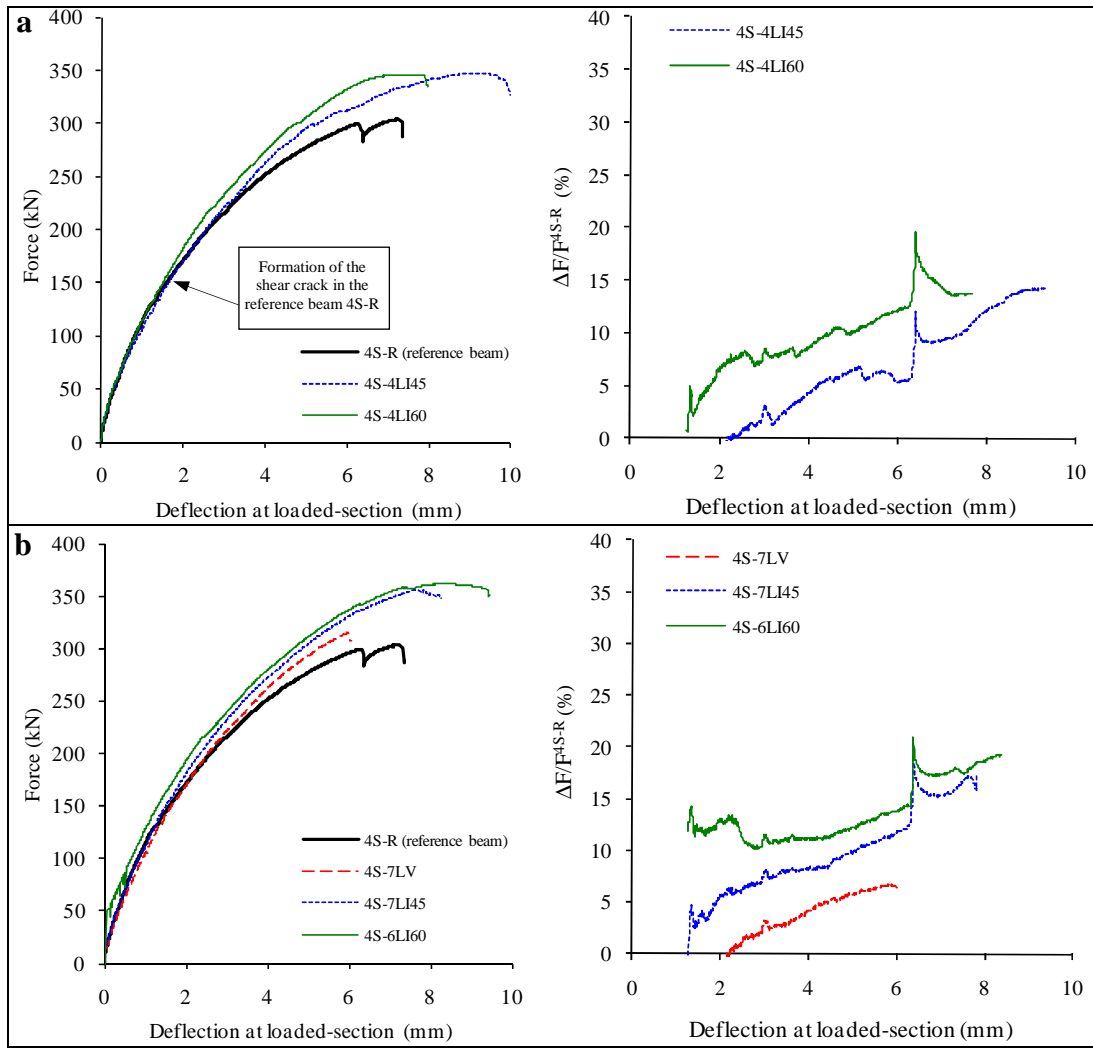


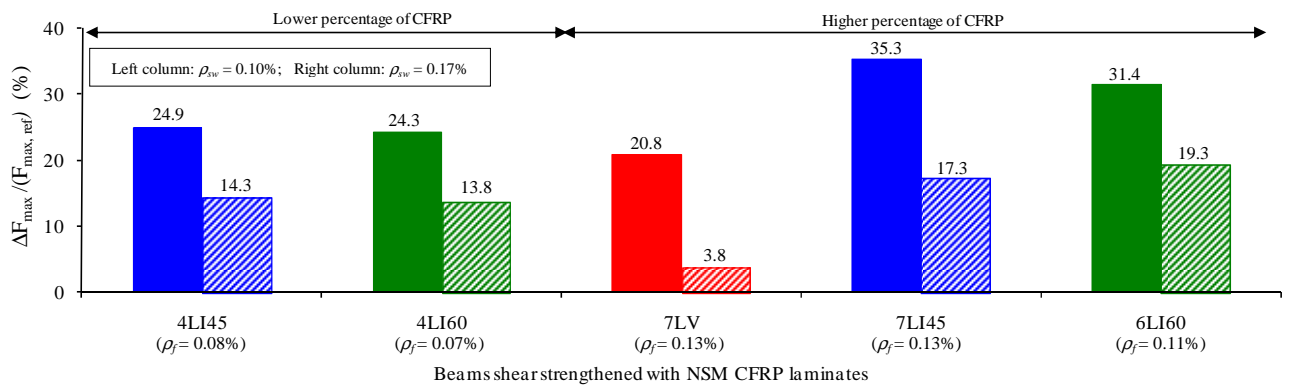
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1 Fig. 5 - Force vs deflection at the loaded-section and $\Delta F/F^{AS-R}$ vs deflection at the loaded-section for the beams with
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4 Beams shear strengthened with NSM CFRP laminates
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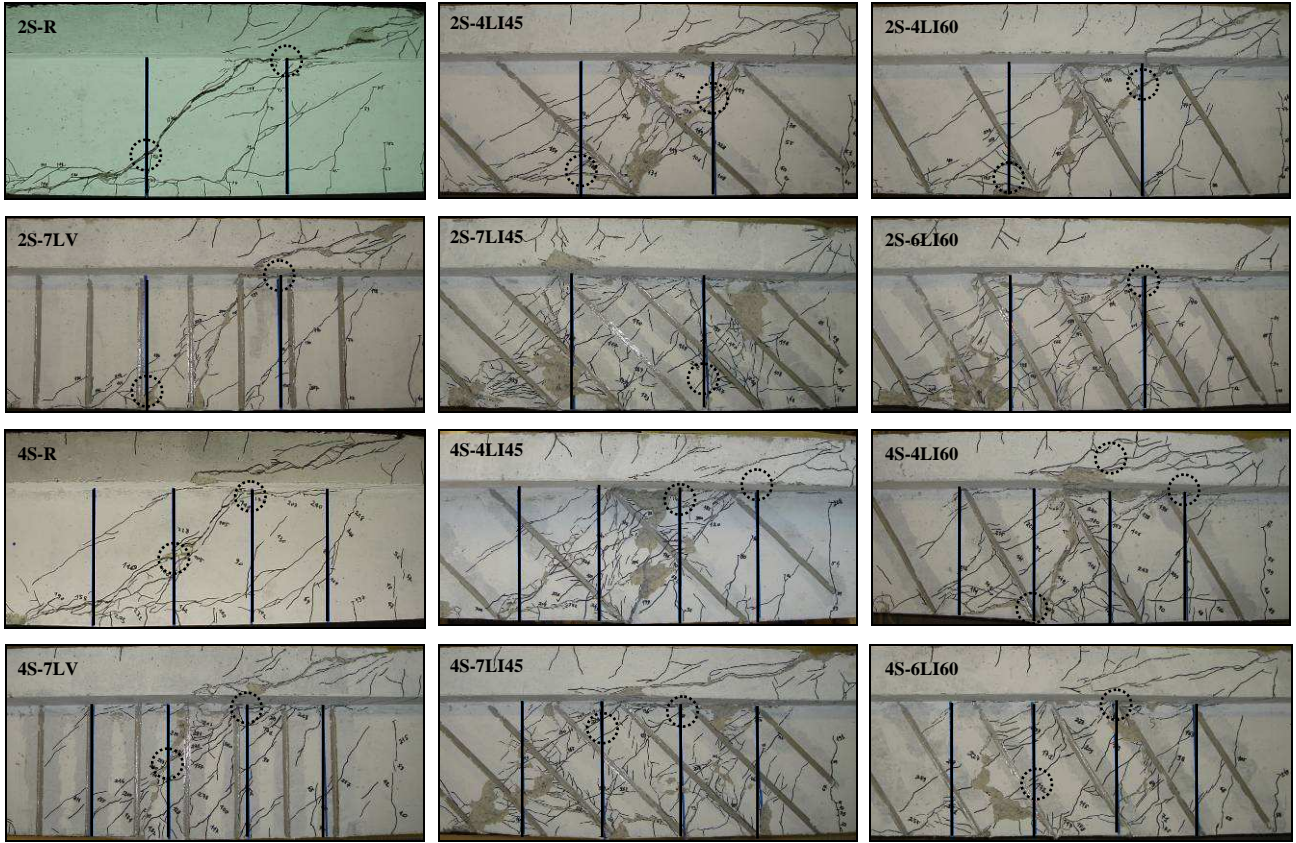


Fig. 7 - Details of the failure zones of the tested beams

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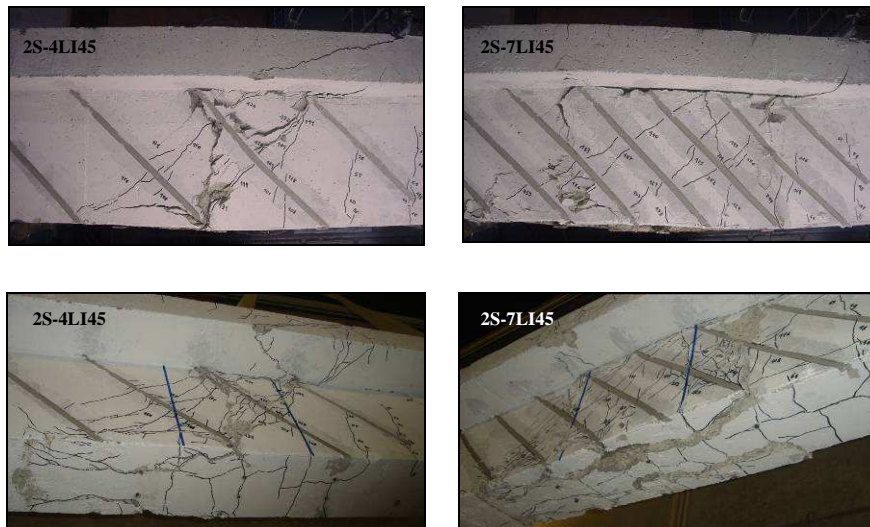
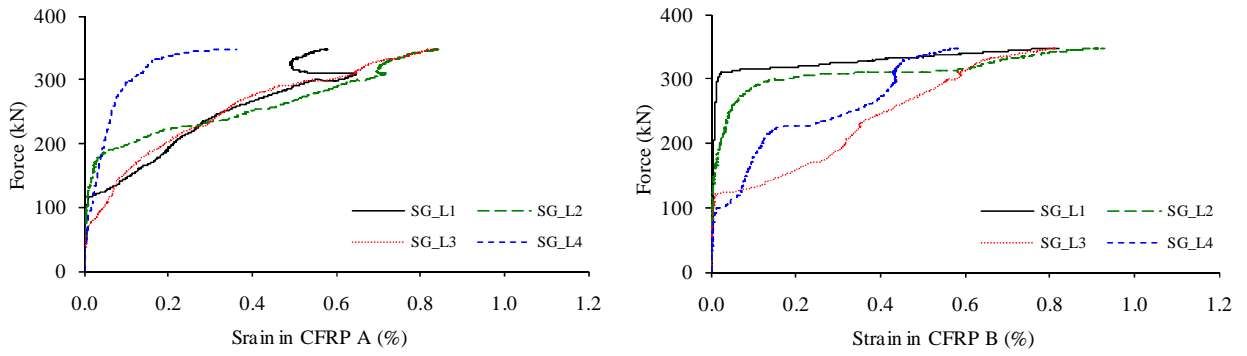
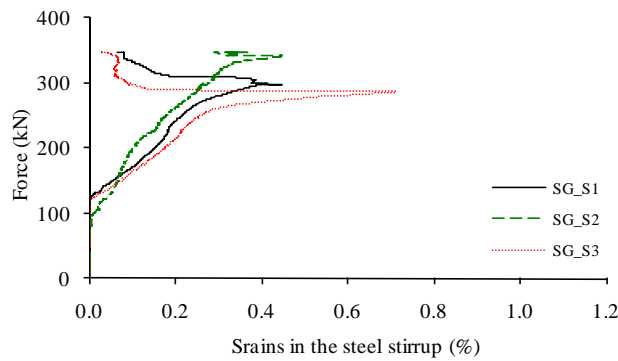


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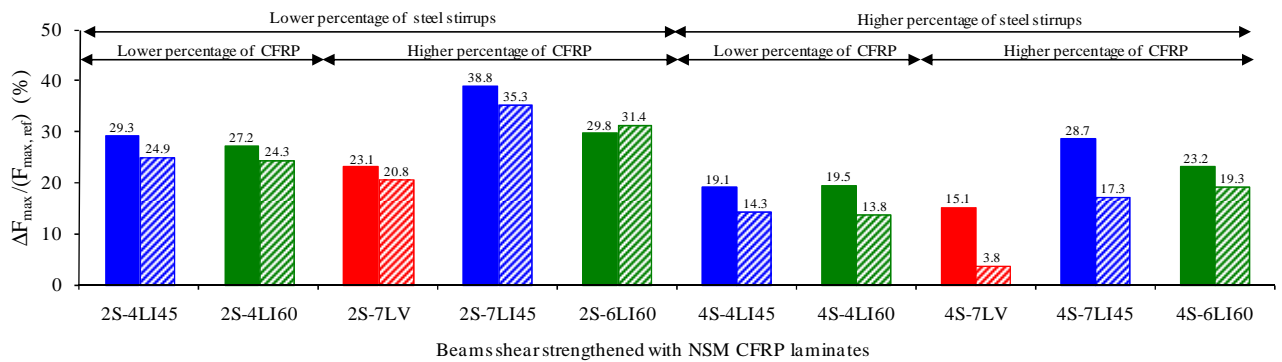
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1 Fig. 9 - Strains in the monitored laminates of the 4S-4LI45 beam



2 Fig. 10 - Strains in the monitored steel stirrup of the 4S-4LI45 beam



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5 Fig. 11 - Influence of the concrete strength in the effectiveness of the NSM shear strengthening technique using
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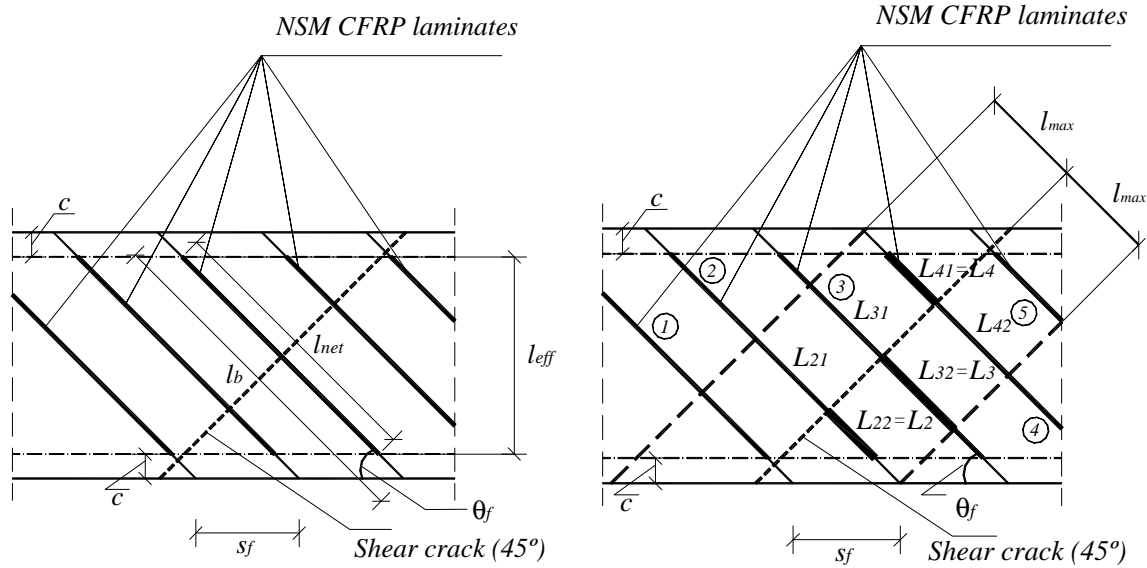


Fig. 12 - Graphical representation of variables used in the formulation by Nanni *et al.* (for this example

$$\sum_i L_i = L_2 + L_3 + L_4$$

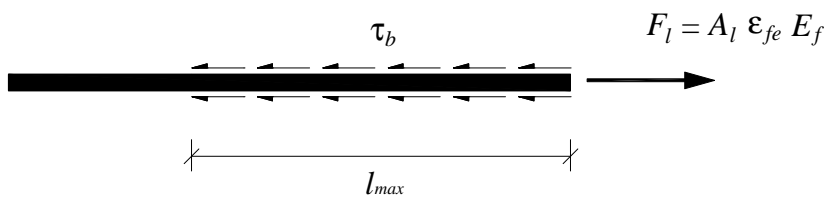


Fig. 13 - Graphical representation of l_{max}

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Table 1 - CFRP shear strengthening configurations of the tested beams

Number of laminates	Angle [θ_f] ($^\circ$) ^a	CFRP spacing [s_f] (mm)	CFRP percentage [ρ_f] (%) ^b	Percentage of steel stirrups [ρ_{sw}]	
				0.10% ^c	0.17% ^d
2×7	90	114	0.13	2S-7LV	4S-7LV
2×4	45	275	0.08	2S-4LI45	4S-4LI45
2×7	45	157	0.13	2S-7LI45	4S-7LI45
2×4	60	243	0.07	2S-4LI60	4S-4LI60
2×6	60	162	0.11	2S-6LI60	4S-6LI60

^a Angle between the CFRP fiber direction and the beam axis; ^b The CFRP percentage was obtained from $\rho_f = (2a_f b_f) / (b_w s_f \sin \theta_f)$ being $a_f = 1.4$ mm and $b_f = 9.5$ mm the dimensions of the laminate cross section, and $b_w = 180$ mm is the beam web width; ^c 2S-R is the reference beam without CFRP (Fig. 2); ^d 4S-R is the reference beam without CFRP (Fig. 2).

Table 2 - Values of the properties of intervening materials

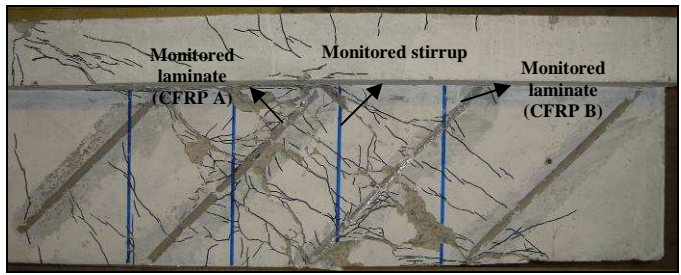
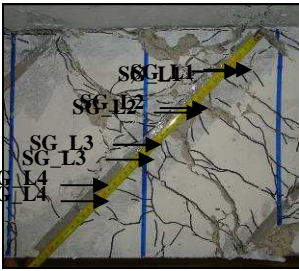
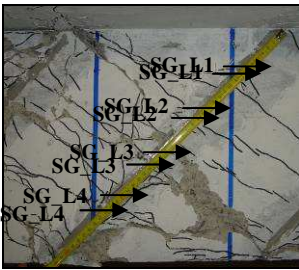
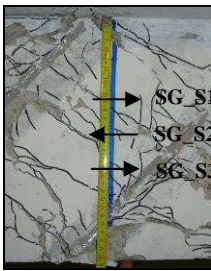
Concrete	Compressive strength [f_{cm}] (MPa)				
	15.9 MPa (at 28 days)		18.6 MPa (at 51 days - age of beam tests)		
Steel	Diameter (mm)	$\phi 6$	$\phi 12$	$\phi 16$	$\phi 32$
	Yield stress [f_{sym}] (MPa)	539	453	429	734
	Tensile strength [f_{sum}] (MPa)	595	581	563	885
CFRP Laminates	Tensile strength [f_{fim}] (MPa)	Young's Modulus [E_{fm}] (GPa)		Maximum strain [ε_{fu}] (%)	
	2847.9	174.3		1.63	

Table 3 - Relevant results in terms of the load capacity up to beam's failure

Beam designation	$(\Delta F / F^{ref})_{max}$ (%)	$u(\Delta F / F^{ref})_{max}$ (mm)	F_{max} (kN)	$\Delta F_{max} / F_{max}^{ref}$ (%)	$u_{F_{max}}$ (mm)	ε_{CFRP}^{max} (%)	$(\varepsilon_{CFRP}^{max})_{med}$ (%)
C-R	-	-	147.0	-	3.44	-	-
2S-R	-	-	226.5	-	5.29	-	-
2S-7LV	26.1	4.48	273.7	20.8	4.55	0.57	0.49
2S-4LI45	24.9	5.79	283.0	24.9	5.79	0.94	0.72
2S-7LI45	38.5	4.50	306.5	35.3	4.79	0.65	0.53
2S-4LI60	27.4	4.45	281.6	24.3	5.57	0.70	0.69
2S-6LI60	31.4	5.84	297.7	31.4	5.84	0.65	0.59
4S-R	-	-	303.8	-	7.20	-	-
4S-7LV	6.7	5.88	315.2	3.8	5.98	0.71	0.41
4S-4LI45	14.3	9.28	347.2	14.3	9.28	0.93	0.88
4S-7LI45	19.3	6.38	356.4	17.3	7.83	0.71	0.60
4S-4LI60	19.6	6.38	345.6	13.8	7.67	0.78	0.75
4S-6LI60	20.9	6.38	362.3	19.3	8.36	0.54	0.52

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Table 4 - Strain variation in monitored laminates and steel stirrup of 4S-4LI45 beam (strain values in %)

Description								
The beam failed by debonding of the CFRP A at $F_{max} = 347.2$ kN.								
CFRP A		SG_L	F =150 kN	F =200 kN	F =250 kN	F =300 kN	F =340 kN	F =347.2 kN
		1	0.11	0.21	0.33	0.61	0.53	0.58
		2	0.02	0.11	0.39	0.68	0.78	0.84
		3	0.09	0.20	0.33	0.58	0.78	0.84
		4	0.03	0.05	0.07	0.11	0.22	0.36
CFRP B		SG_L	F =150 kN	F =200 kN	F =250 kN	F =300 kN	F =340 kN	F =347.2 kN
		1	0.00	0.00	0.01	0.02	0.62	0.82
		2	0.10	0.02	0.05	0.18	0.79	0.93
		3	0.17	0.32	0.42	0.58	0.74	0.81
		4	0.08	0.12	0.33	0.43	0.52	0.58
Steel stirrup		SG_S	F =150 kN	F =200 kN	F =250 kN	F =300 kN	F =340 kN	F =347.2 kN
		1 ^a	0.06 (0.14)	0.15 (0.23)	0.22 (0.27)	0.38 (0.35)	0.08	0.06
		2 ^a	0.07 (0.19)	0.10 (0.45)	0.18 (0.68)	0.28 (0.84)	0.43	0.37
		3 ^b	0.08	0.18	0.26	0.09	0.06	0.03

Note: To localize the SGs applied in the two arms of the steel stirrup, the arrow which points to the left indicates the SG applied in the arm at the opposite side of the one represented in the Figure. ^a Values in brackets are those recorded in the 4S-R beam at $F_{max}=303.8$ kN. ^b This SG did not work in the reference beam 4S-R.

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1 Table 5 - Influence of the concrete strength in the effectiveness of the NSM shear strengthening technique with
 2 CFRP laminates

CFRP		$f_{cm} = 39.7 \text{ MPa}$				$f_{cm} = 18.6 \text{ MPa}$			
Percentage (%)	Angle (°)	Beams	$\Delta F_{max} / F_{max}^{ref}$	$u_{F_{max}} / u_{F_{max}}^{ref}$	ε_{CFRP}^{max}	Beams	$\Delta F_{max} / F_{max}^{ref}$	$u_{F_{max}} / u_{F_{max}}^{ref}$	ε_{CFRP}^{max}
			(%)	(%)	(%) ^a		(%)	(%) ^b	(%)
0.13	90	2S-7LV	23.1	21.9	0.77	2S-7LV	20.8	-14.0	0.57
0.08	45	2S-4LI45	29.3	9.7	1.08	2S-4LI45	24.9	9.5	0.94
0.07	60	2S-4LI60	27.2	17.3	0.99	2S-4LI60	24.3	5.3	0.70
0.13	45	2S-7LI45	38.8	34.9	0.85	2S-7LI45	35.3	-9.5	0.65
0.11	60	2S-6LI60	29.8	33.8	0.99	2S-6LI60	31.4	10.4	0.65
0.13	90	4S-7LV	15.1	56.0	0.91	4S-7LV	3.8	-16.9	0.71
0.08	45	4S-4LI45	19.1	26.9	0.79	4S-4LI45	14.3	28.9	0.93
0.07	60	4S-4LI60	19.5	10.6	0.94	4S-4LI60	13.8	6.5	0.78
0.13	45	4S-7LI45	28.7	32.2	0.82	4S-7LI45	17.3	8.8	0.71
0.11	60	4S-6LI60	23.2	17.0	0.87	4S-6LI60	19.3	16.1	0.54

^a For the beams 4S-7LV, 4S-4LI45, 4S-4LI60, 4S-7LI45 and 4S-6LI60 only one laminate has been monitored. ^b In the beams with negative value the deflection at the loaded section for the maximum load was lower than that of observed in the reference beam.

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Table 6 - Analytical vs experimental results for the tested beams with NSM CFRP laminates

Beam designation	Experimental	Formulation by Nanni <i>et al.</i> [11]	
	$V_f^{exp.}$ (kN)	V_{fd}^{ana} (kN)	$V_f^{exp.} / V_{fd}^{ana}$
2S-7LV	28.3	29.9	0.95
2S-4LI45	33.9	14.0	2.42
2S-7LI45	48.0	34.3	1.40
2S-4LI60	33.1	17.1	1.94
2S-6LI60	42.7	34.2	1.25
4S-7LV	6.8	29.9	0.23
4S-4LI45	26.0	14.0	1.86
4S-7LI45	31.6	34.3	0.92
4S-4LI60	25.1	17.1	1.47
4S-6LI60	35.1	34.2	1.03

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